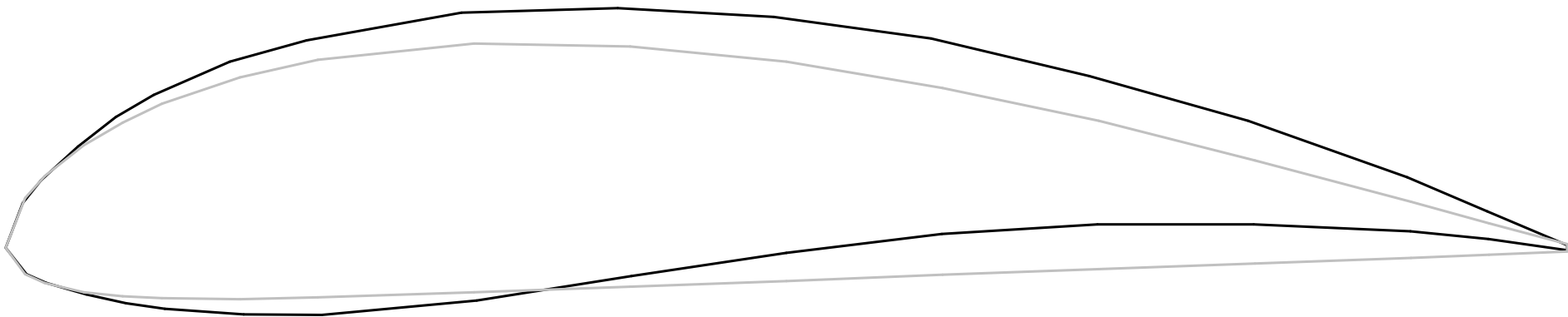


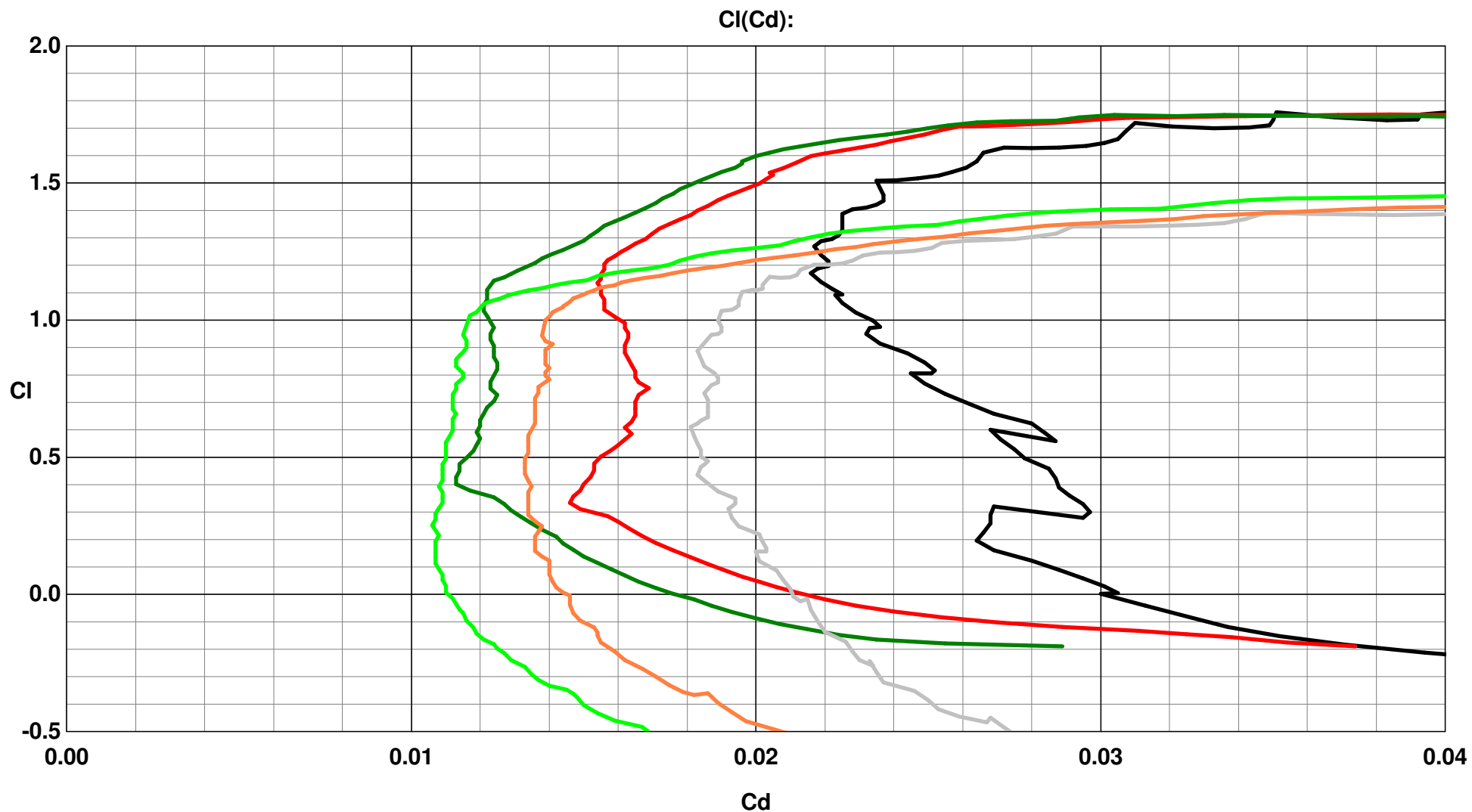
Airfoils in comparison :

GOE 222 (MVA H,33) = _____

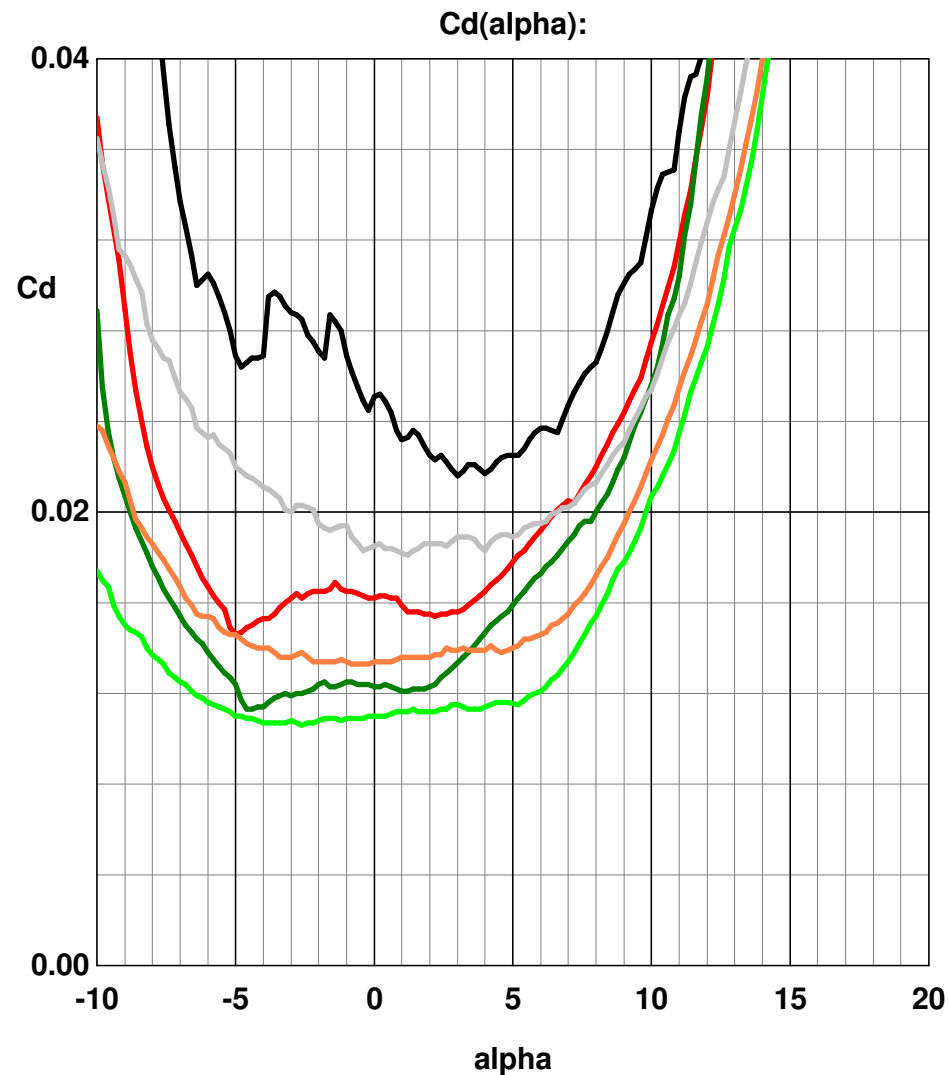
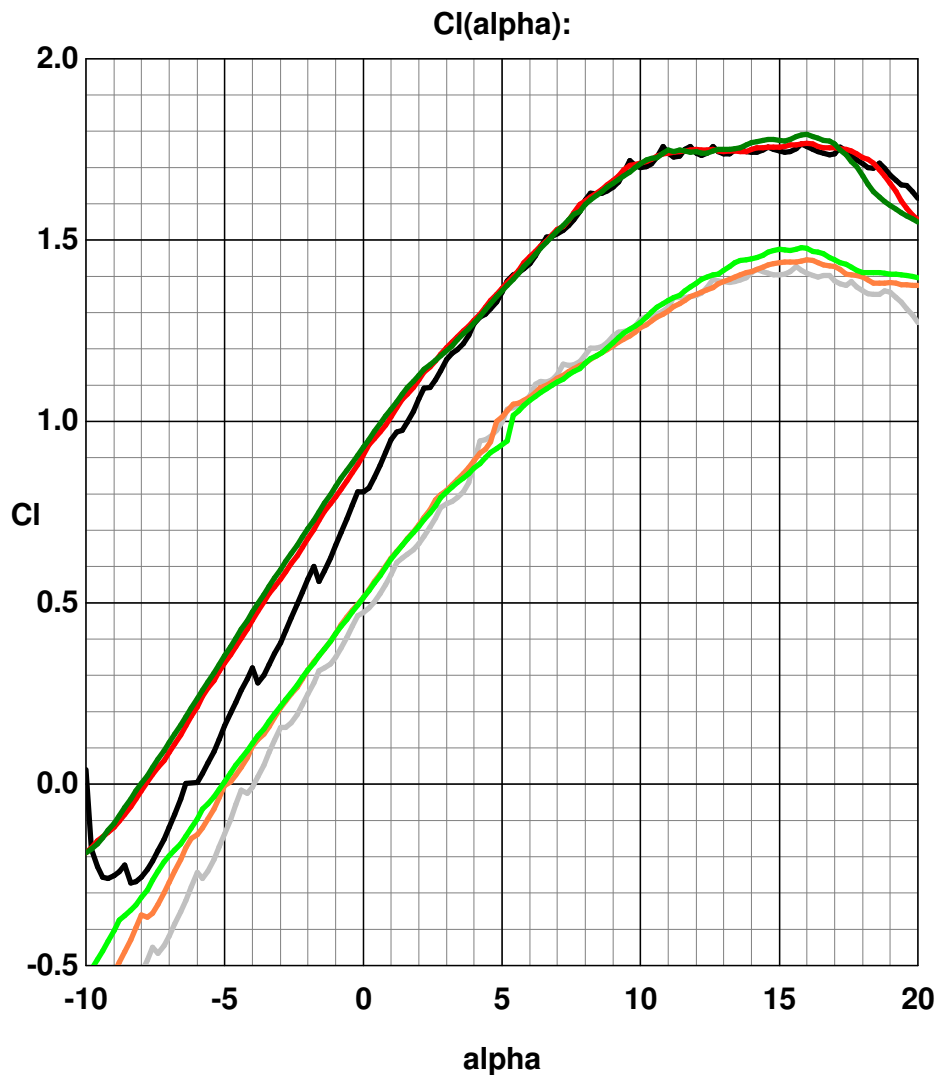
GOE 624 = _____



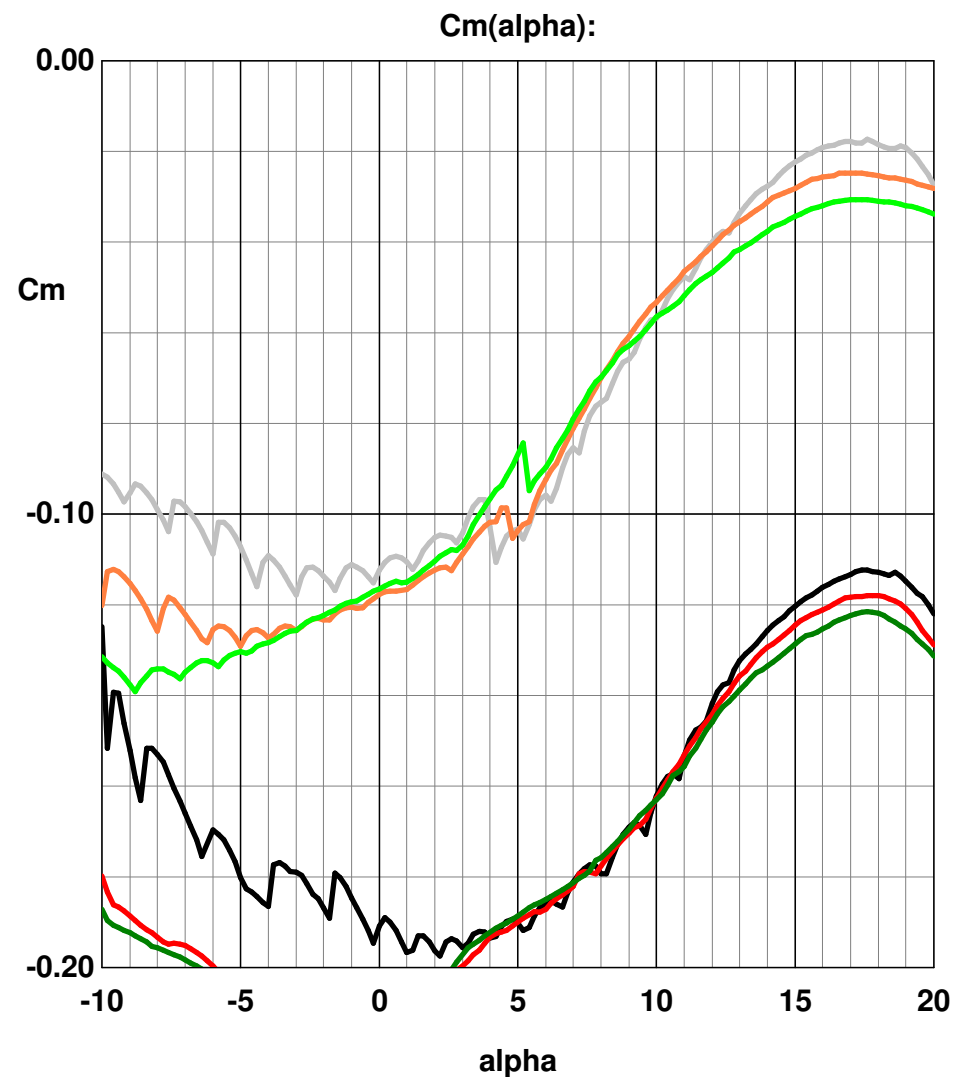
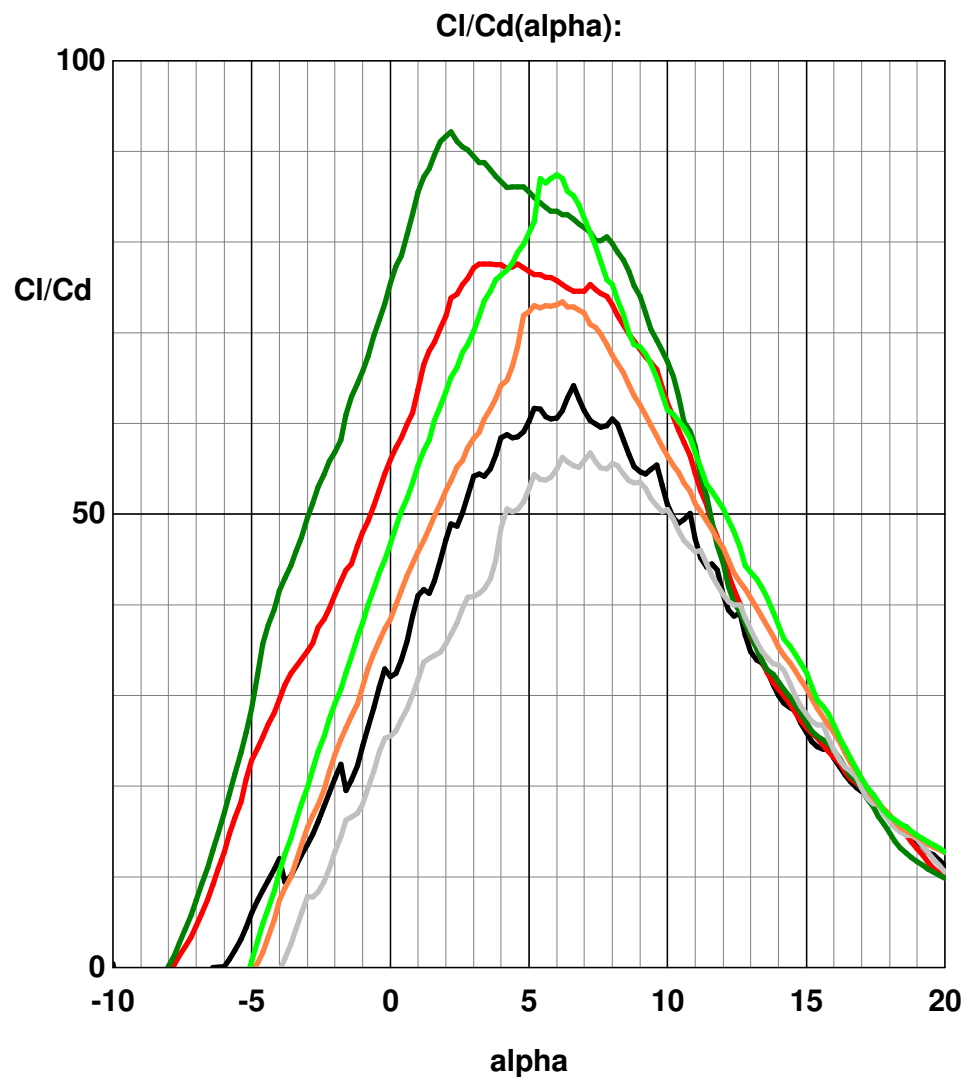
- 'GOE 222 (MVA H,33)' at 150000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 222 (MVA H,33)' at 250000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 222 (MVA H,33)' at 400000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 646' at 150000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 646' at 250000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 646' at 400000 Re - Mach=0.0000 - NCrit=9.00



- 'GOE 222 (MVA H,33)' at 150000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 222 (MVA H,33)' at 250000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 222 (MVA H,33)' at 400000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 646' at 150000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 646' at 250000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 646' at 400000 Re - Mach=0.0000 - NCrit=9.00

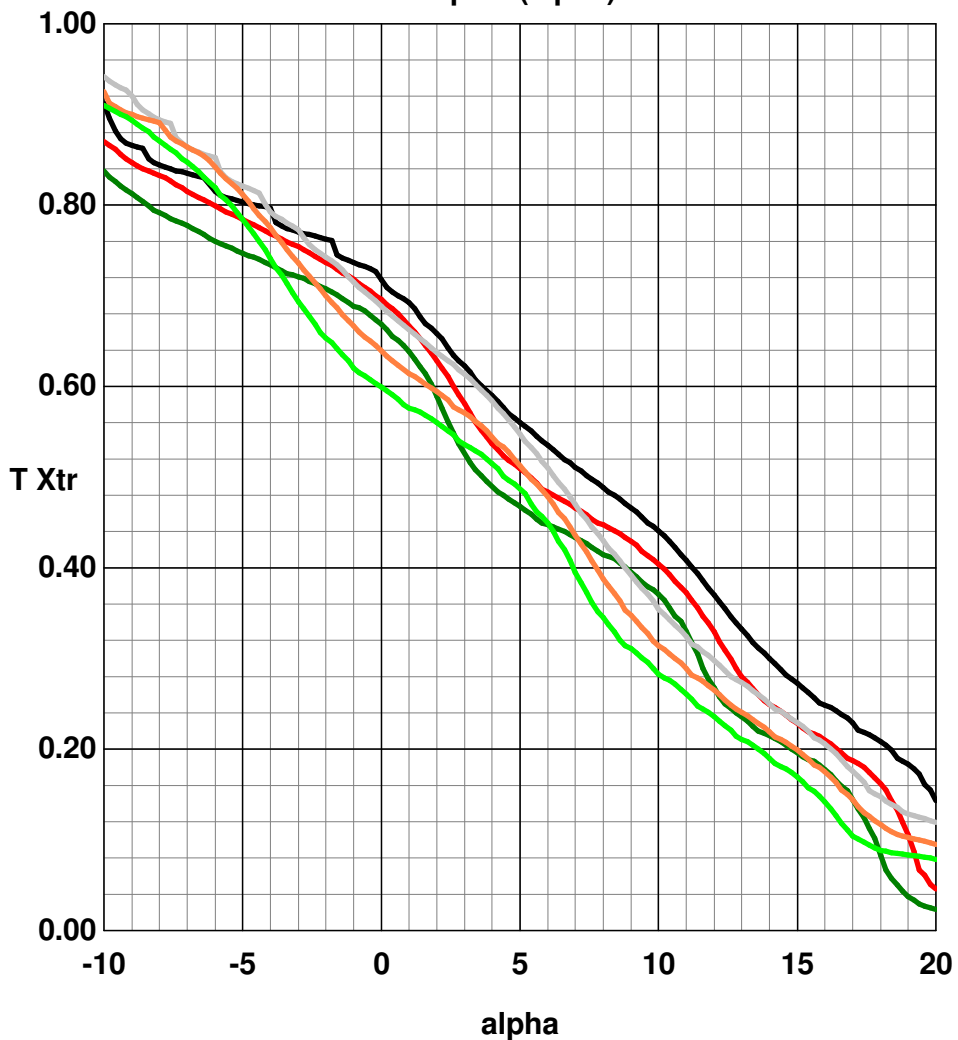


- 'GOE 222 (MVA H,33)' at 150000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 222 (MVA H,33)' at 250000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 222 (MVA H,33)' at 400000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 646' at 150000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 646' at 250000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 646' at 400000 Re - Mach=0.0000 - NCrit=9.00

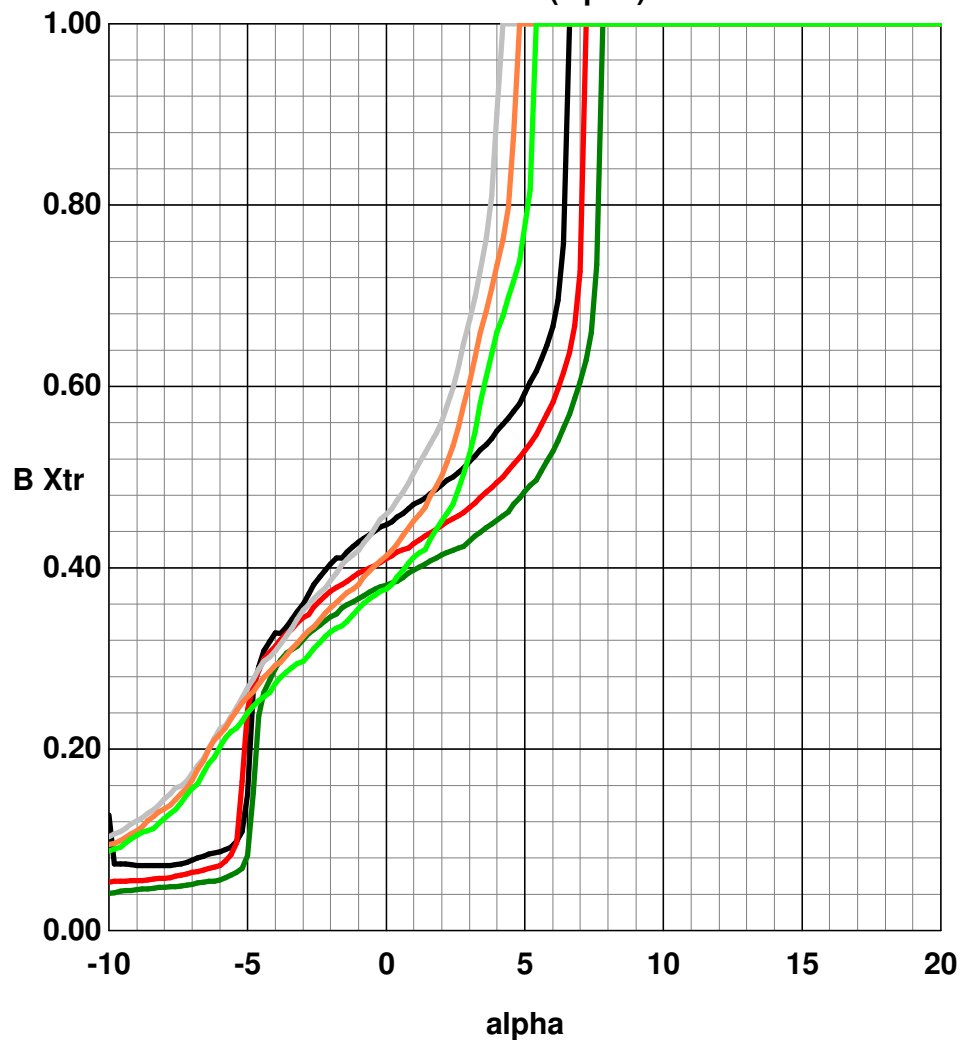


- 'GOE 222 (MVA H,33)' at 150000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 222 (MVA H,33)' at 250000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 222 (MVA H,33)' at 400000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 646' at 150000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 646' at 250000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 646' at 400000 Re - Mach=0.0000 - NCrit=9.00

Top Xtr(alpha):



Bottom Xtr(alpha):



- 'GOE 222 (MVA H,33)' at 150000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 222 (MVA H,33)' at 250000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 222 (MVA H,33)' at 400000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 646' at 150000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 646' at 250000 Re - Mach=0.0000 - NCrit=9.00
- 'GOE 646' at 400000 Re - Mach=0.0000 - NCrit=9.00

Power Factor(alpha) = $Cl^{1.5}/Cd(alpha)$:

